

The background of the slide features a blue space-themed grid with white stars. Three gold-colored satellite-like spacecraft are positioned at different points. Two smaller spacecraft at the top left and top right are connected by two thick red laser beams that converge towards the center. A larger spacecraft at the bottom right is also connected to the same central area by a red laser beam. The overall scene represents the LISA (Laser Interferometer Space Antenna) mission.

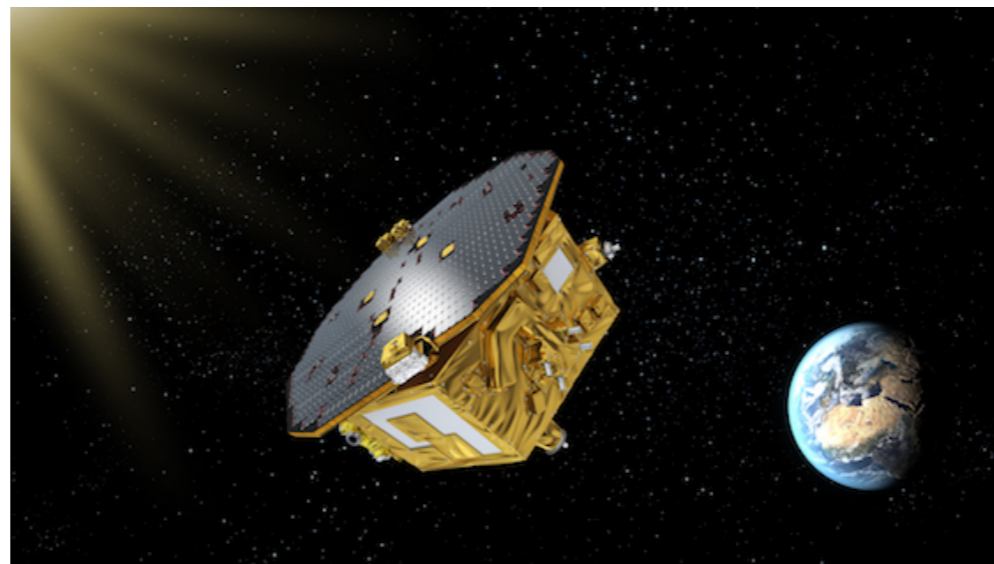
Status of the LISA Data and Diagnostics Subsystem

Miquel Nofrarias

Iberian Gravitational Wave Meeting, June 10th 2021

Organisation

- Funding **approved** for the period 2020-23
- Focused on academic activities
 - *Industrial contribution derived to PRODEX call*
- More groups involved at the IEEC institute, both in experimental and theoretical fronts
 - *Team: 12 staff researchers (2 Science +10 Instrument), 2 staff managers, 5 senior engineers, 4 post-docs, 3 PhD students*
 - *Extensive experience in space instrumentation*



Spanish Mission Lead

Carlos Sopena (IEEC-CSIC) – PI

- LISA Science Study Team
- Consortium Board (Spanish representative)

Miquel Nofrarias (IEEC-CSIC) – co-PI

- LISA System Engineering Office (SEO)
- LISA Instrument Group (Spanish representative)



LISA Science Group (LSG)

C. F. Sopena (IEEC-CSIC) - Spanish lead

C. Giardini (IEEC-UB)

PhD Student(s):

D. Cruces (IEEC-UB)

I. Martin (IEEC-CSIC)

Management Team

Josep Colomé (IEEC-CSIC) – NPM

- National Project Management Board

Alberto García (IEEC)

**INDUSTRIAL
PARTNER(S)**

LISA Instrument Group (LIG)

Data and Diagnostics Subsystem (DDS) M. Nofrarias – Spanish lead

Diagnostics Team

IEEC-CSIC

C. Sierra

D. Roma

JP. López

IEEC-UPC

J. Ramos

M. Domínguez

V. Jiménez

A. Orpella

J. Pons

IEEC-UB

A. Herms

D. Gascón

F. Salvat

E. Graugés

A. Arán

D. Sánchez

**Instrument Control
Computer Team**

IEEC-CSIC

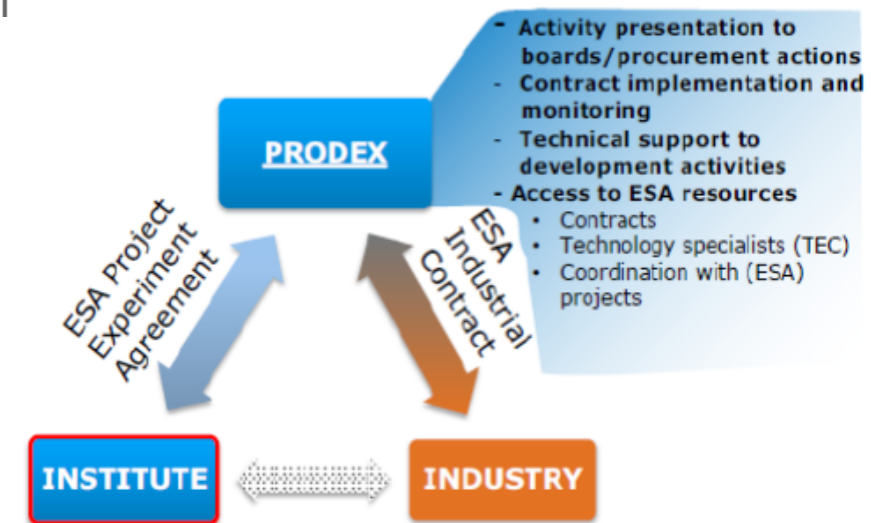
C. Sierra

D. Roma

V. Martin

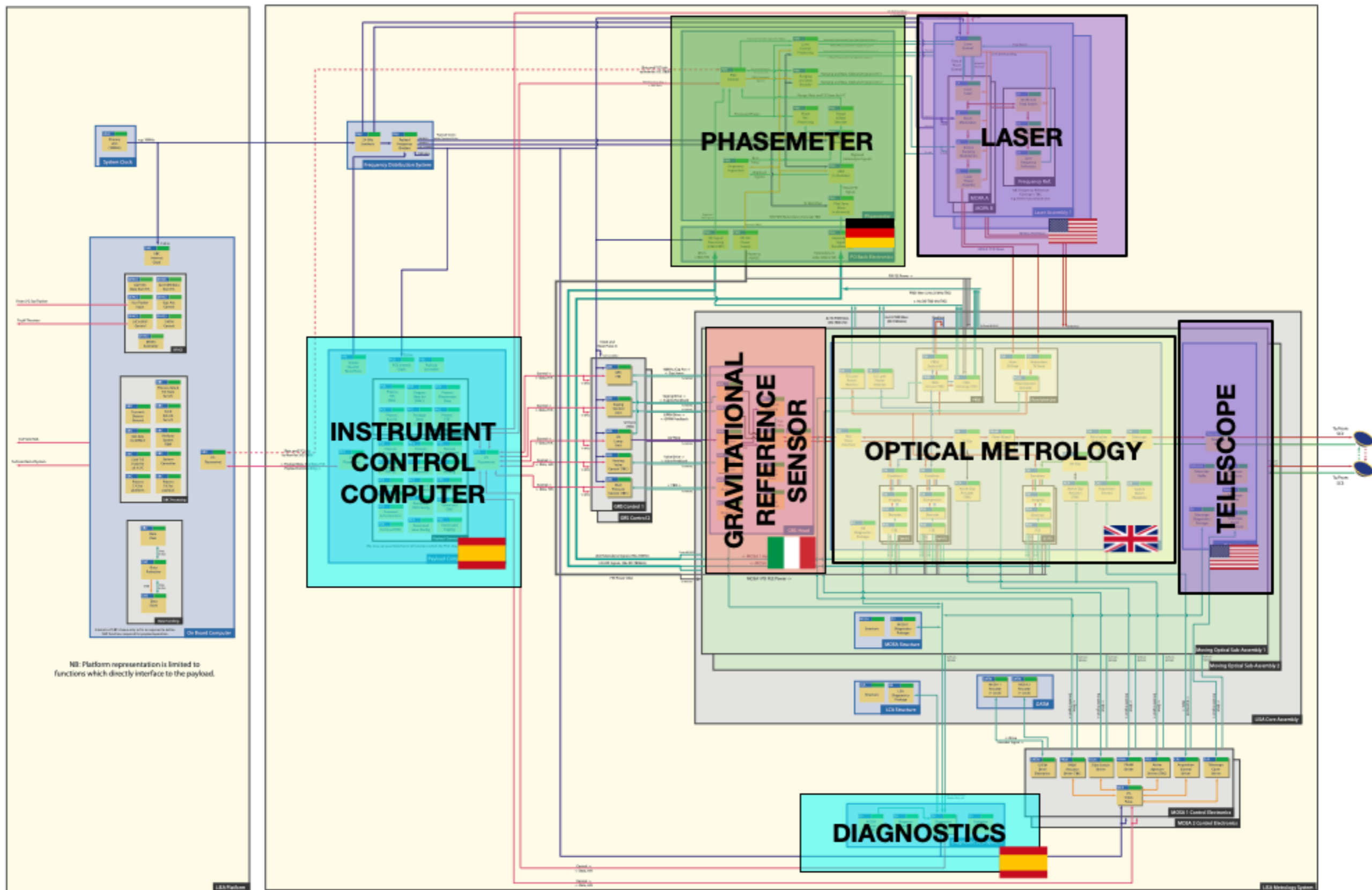
Organisation - PRODEX

- New instrument for space missions: PRODEX
 - Approved in the last CM19: 30 M€
 - Complement the research program and coordinate the academic and industrial activities
 - **Focused on the industrial contributions**
 - Increase the stability of the Spanish contribution to the scientific missions
- Presented LISA phase A study in the first call (2021)
 - Proposal lead by Sener Aeroespacial.
 - *IEEC key partner*
 - Type A: phase A
 - Duration: 12 months
 - Financial proposal: 299 k
- Scope
 - Boost Spanish contribution to the Diagnostics Subsystem, as the core element of the LISA Payload lead by Sener Aeroespacial.
 - Key technologies
 - A feasible concept for the **sensors front-end electronics** to achieve the needed low-noise performance while minimizing the thermal impact on the MOSA
 - An architecture definition for the Payload **Data Processing Function** (SW, HW, FW) fulfilling the overall payload needs, and the data processing requirements towards the formation flying management (3 satellites constellation)
 - Intended to consolidate the **Spanish Industry contribution** to an L-Mission as LISA technologies
- Letter of support from LISA Consortium



SPACECRAFT

INSTRUMENT



Note: Platform representation is limited to functions which directly interface to the payload.

Change Log

2017-09-15: Initial release version
 2017-08-15: Revised instrument layout design
 2017-08-10: Change following JSC-PT2 contract award and
 2017-08-05: Revised instrument layout design and layout for
 2017-08-01: Change following contract award and layout for
 2017-07-25: Add final additional change of instrument layout and
 2017-07-20: Final version

Key

- Power:**
 - Red: Main Power
 - Blue: System Power
 - Green: Standby Power
- Optical:**
 - Red: Laser Light
 - Blue: Starlight
 - Green: Reference Light
- Data and Control:**
 - Blue: Control Signals
 - Green: Data Signals
 - Red: Telemetry
- Functional Coupling:**
 - Blue: Functional Coupling
 - Green: Sub-System
- Standard Function used for Block:**
 - Blue: Standard Function used for Block
 - Green: Function under Development with Internal Reference

Hardware implementation scheme

- Consolidation of the Spanish role (at the current stage)

The slide, titled "LISA Implementation Scheme (Member States)", features the ESA logo in the top right corner. It lists three main implementation areas:

- Implementation scheme (Hardware contributions) for B1 and beyond under discussion with Member States
- In addition, provision of performance modelling, operations and calibration schemes, L0-L1 processing support to ESA
- L2 and L3 Data Processing (Science Data Processing Centres, led by France)

Below these points, hardware contributions are organized into four columns, each with a color-coded header box and a small flag icon:

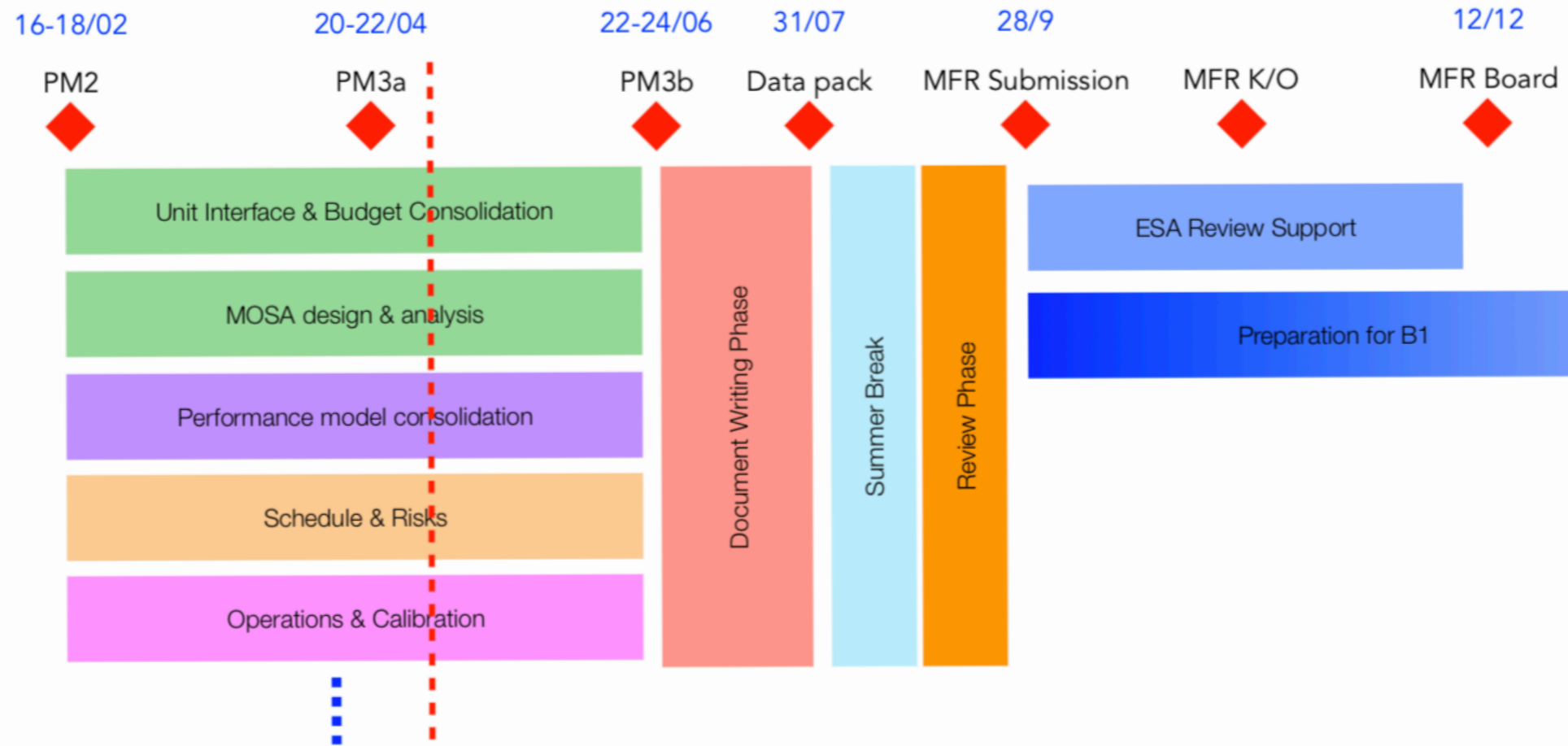
- Green column:** Gravitational Reference System (Italy), GRS Head (Italy), GRS FEE (Switzerland), GRS MCU (Italy), and CMD (USA).
- Red column:** Interferometric Detection System (Germany), Optical Bench (UK), ePMS & PSCU (Germany), OB Mechanisms (France, Belgium, Netherlands), and Photoreceivers (France, Germany, Japan).
- Yellow column:** Optical GSE (France).
- Grey column:** Data and Diagnostics (Spain), ICC (Spain), and Sensors and DSCU (Spain).

A "Preliminary" watermark is visible across the center of the slide. At the bottom, a row of flags represents the LISA Consortium members, and the text "THE EUROPEAN SPACE AGENCY" is on the right.

M. Gehler (LISA Study Project Manager) slide, LISA Consortium, June 2021.

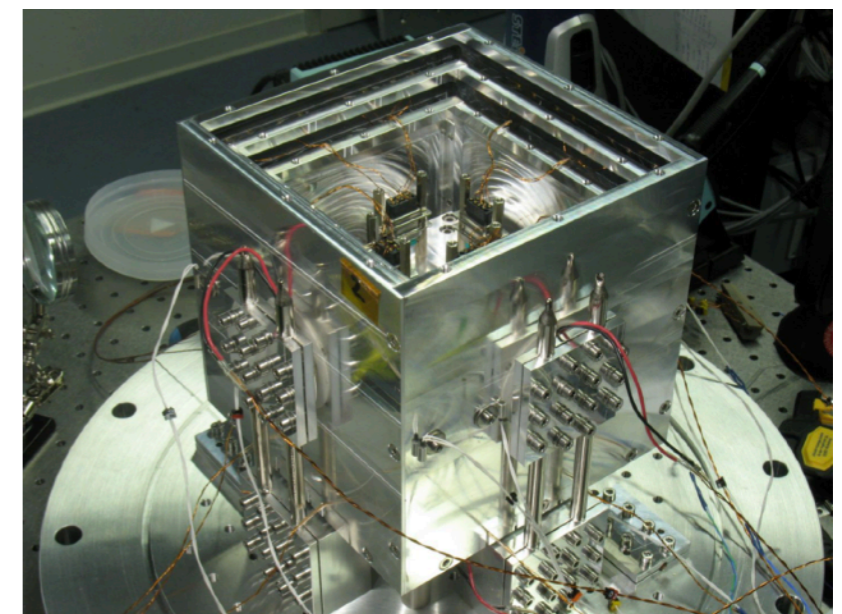
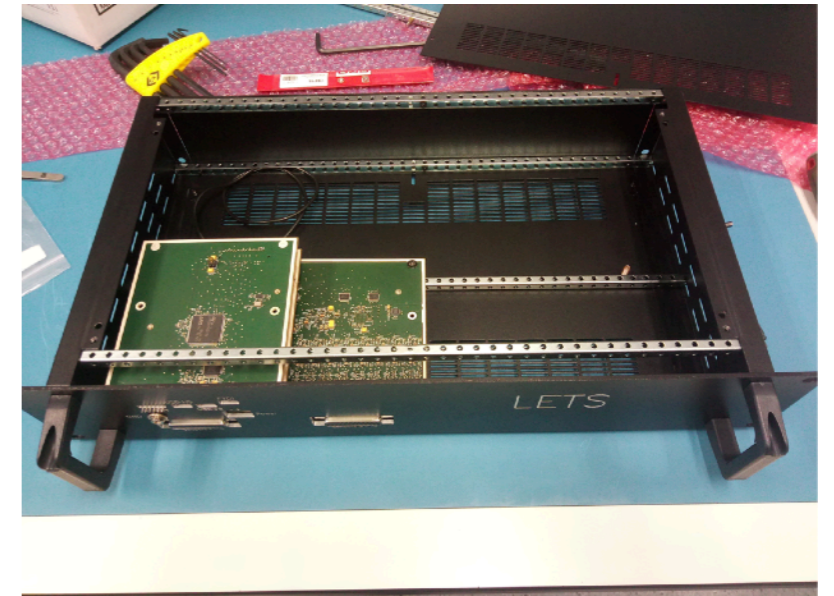
LISA Phase A schedule

Phase A2 Milestones



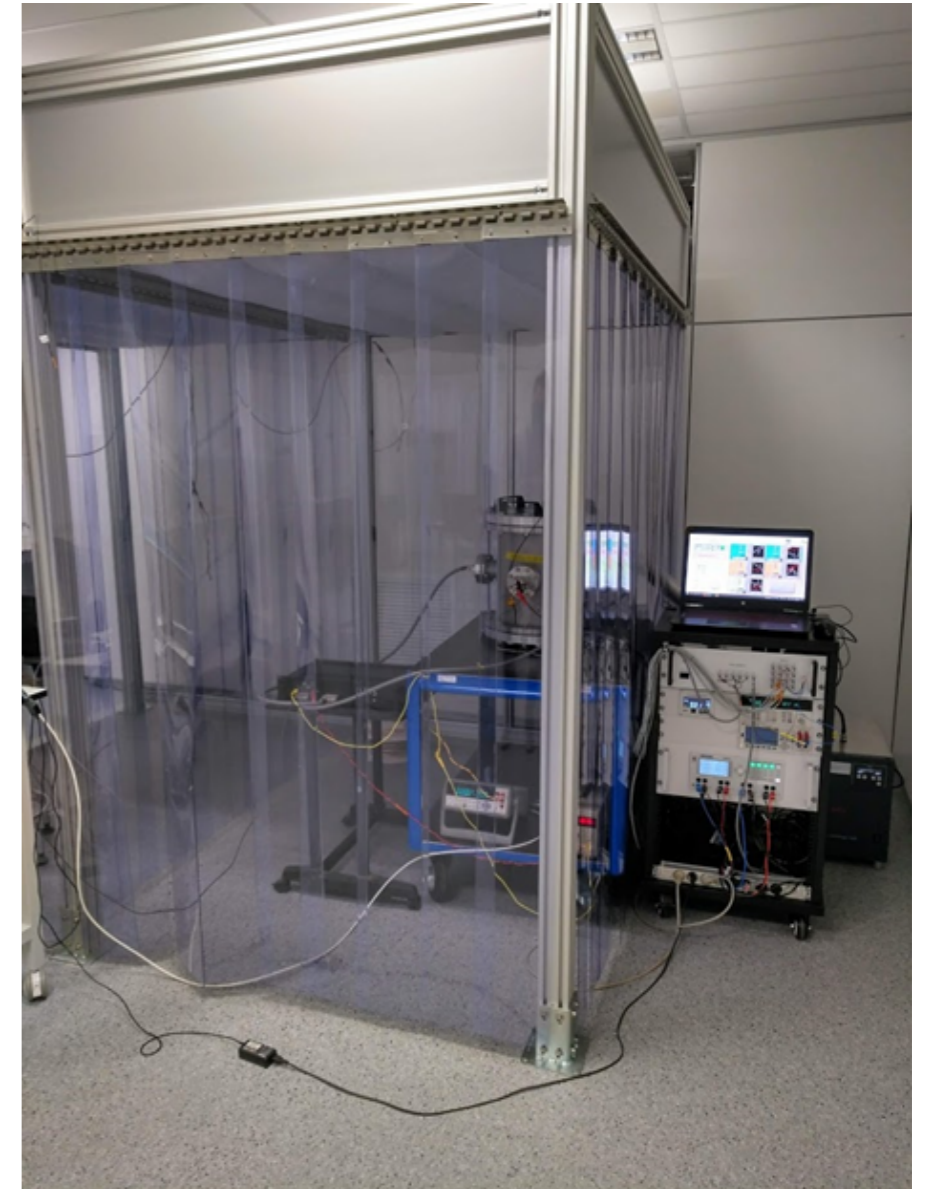
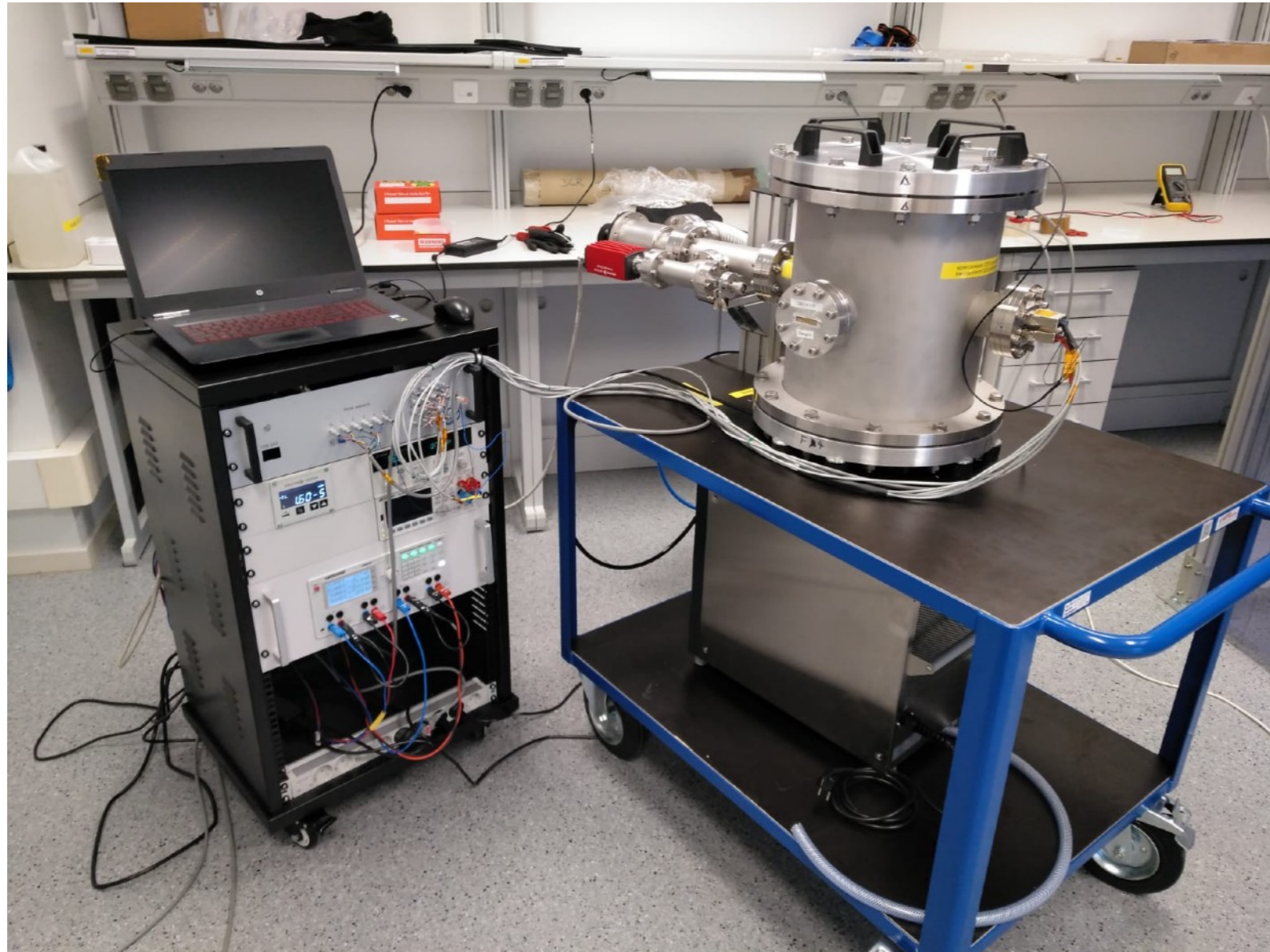
LISA – temperature diagnostics subsystem

- Developing LISA temperature diagnostics subsystem under ‘**Enhanced temperature measurement for LISA**’ (LETS) ESA contract
 - Team: IEEC (ES), DLR (DE), SENER (ES)
 - Duration 18 months (delayed due to COVID)
- The objective is the design of a prototype temperature subsystem for LISA (TRL4)
 - Increasing 1 order of magnitude performance: $1\mu\text{K}/\sqrt{\text{Hz}}$ down to 1mHz
- Two main components:
 - **Front-end electronics** composed by Analog Front-end Board (AFB), Power Distributing Board (PDB) and Digital Processing Board (DPB)
 - **Ultra-stable test bench** (DLR) composed by concentric Al thermal shield layers inside vacuum tank. Peltier elements for active control
- Current status:
 - Aiming for Final Presentation on June 30th



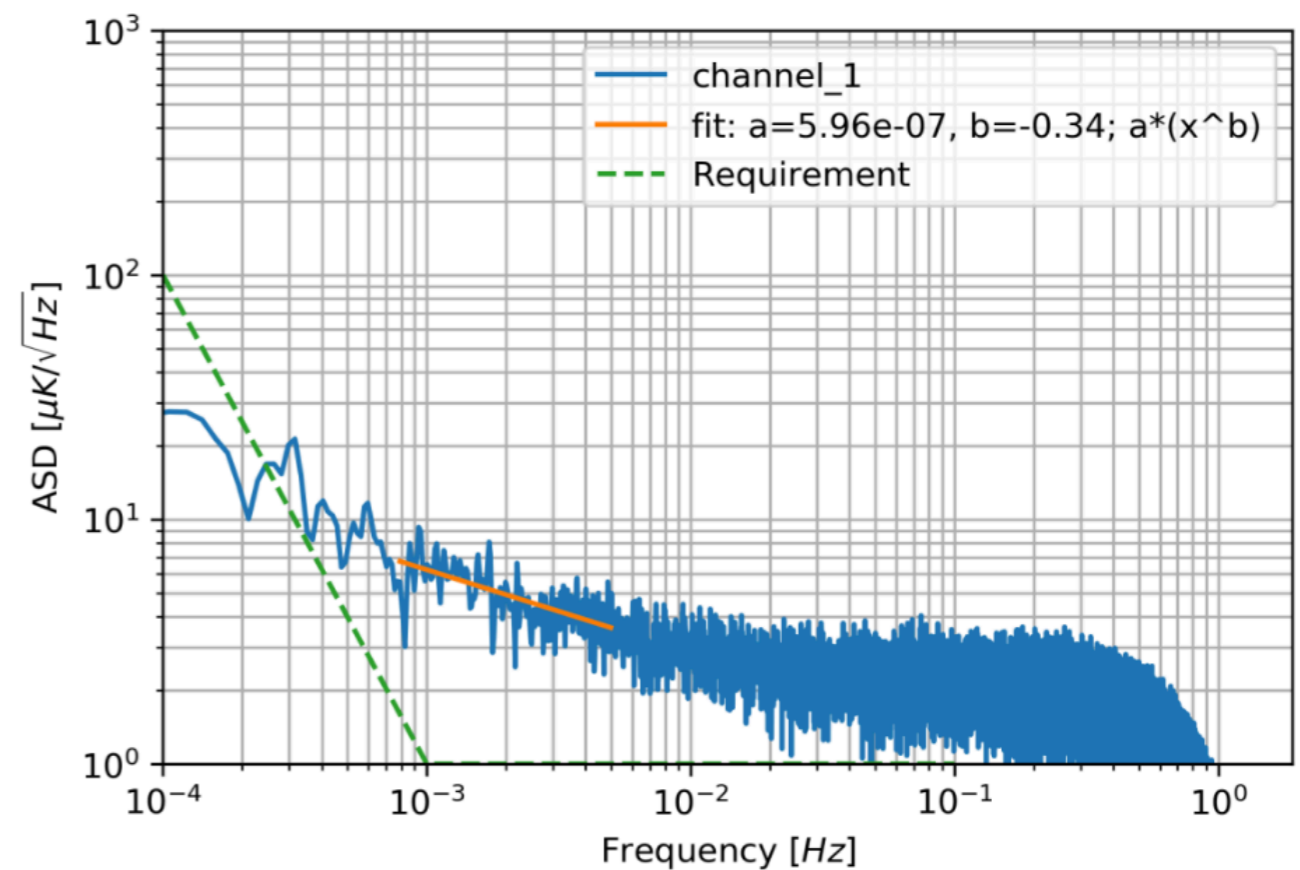
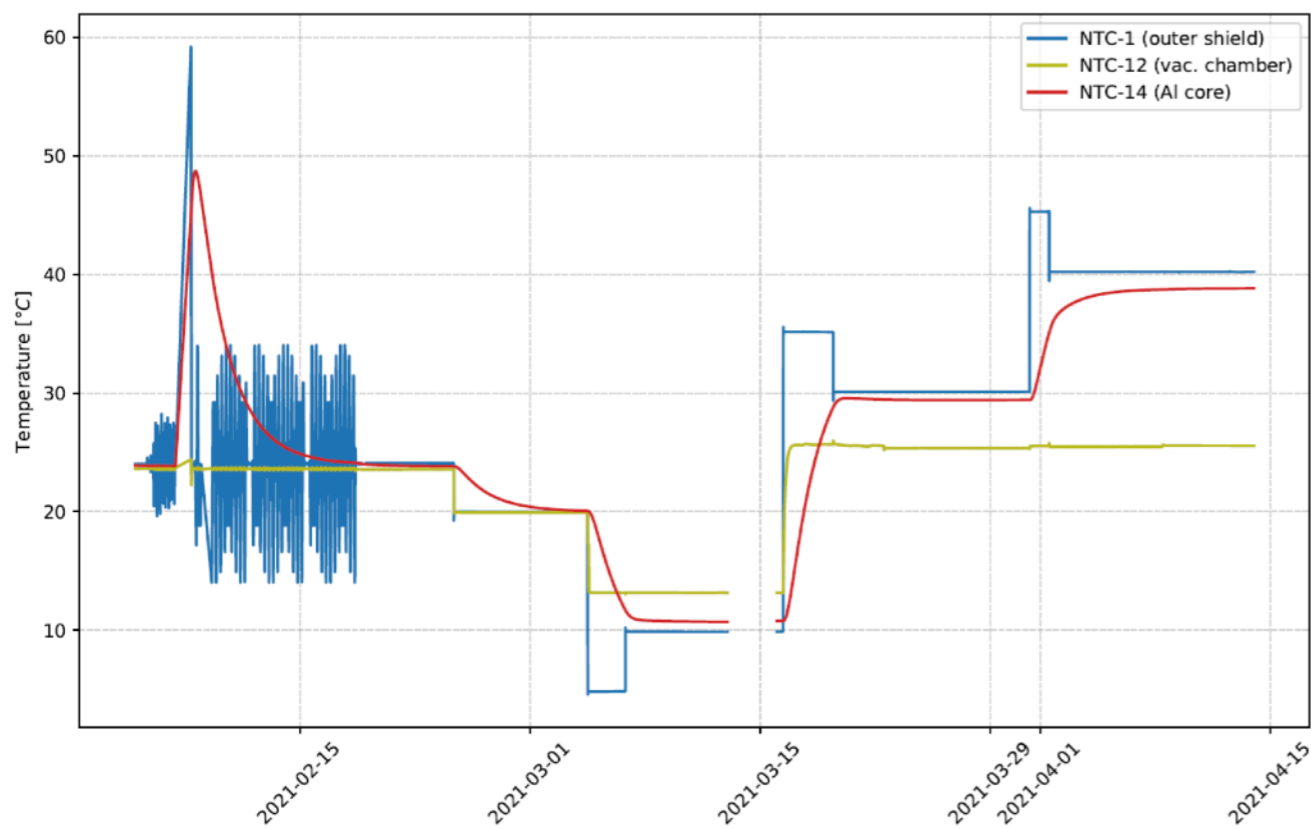
LISA – temperature diagnostics

- Setup running at ICE, inside temperature control cabine ($\pm 5\text{mK}$ stability)



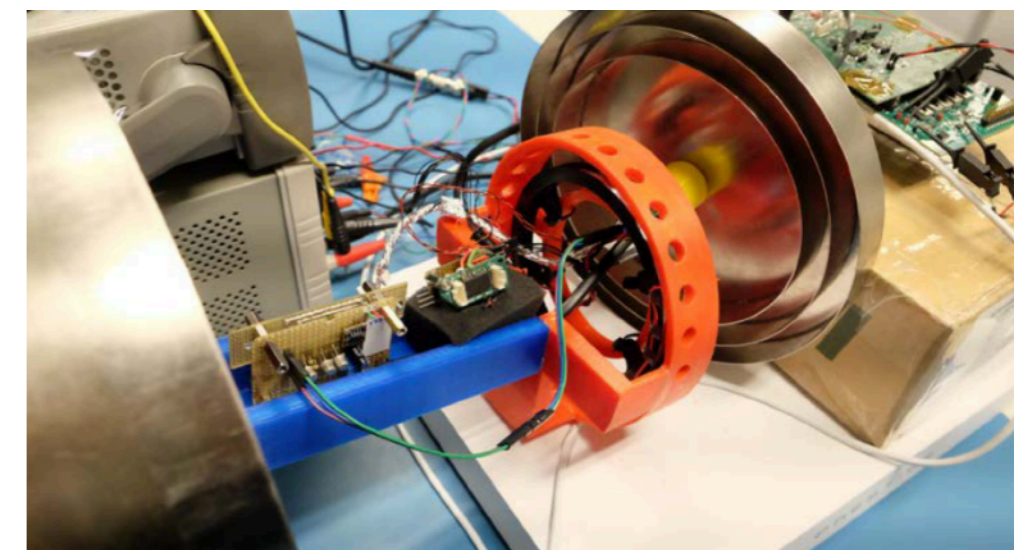
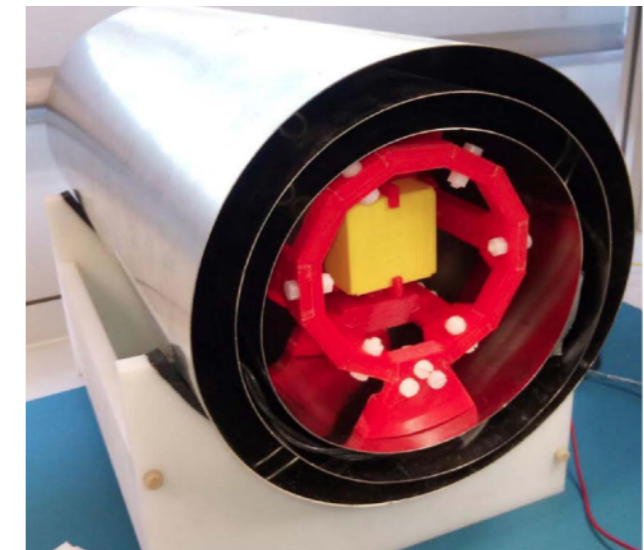
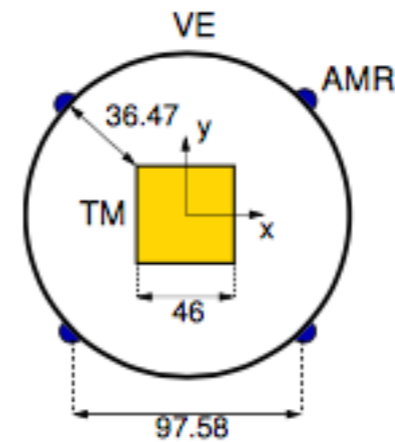
LISA – temperature diagnostics

- Noise runs running since Feb. '21.
 - Long runs required to reach 0.1mHz regime
- After extensive testing, we've been able to track down several noise contributions. Close to the requirement. Floor noise 2-3 $\mu\text{K}/\sqrt{\text{Hz}}$, still some excess noise at 1mHz
- Remaining excess noise being investigated (harness, grounding, interferences...)

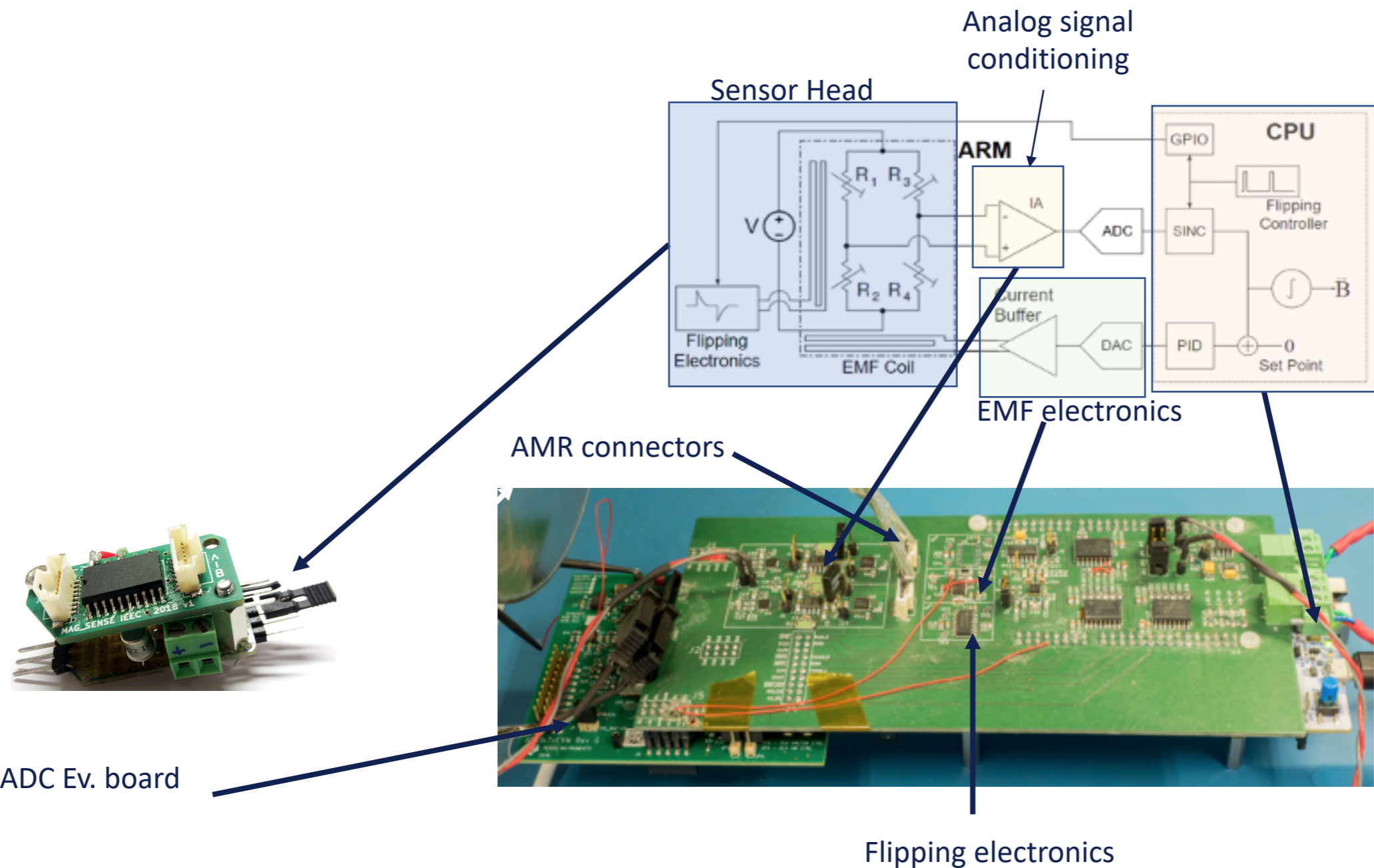


LISA – magnetic diagnostics (AMRs)

- We have developed an improved magnetic diagnostic system more compact and avoiding back-action problems
 - based on **Anisotropic magneto-resistors (AMR)**, solid-state, low noise magnetic sensors.
 - AMR is a compact, low-noise with no magnetic back-action
 - A solution with 8 AMRs would reduce error to 1%
- Setup for testing is composed by
 - three concentric **mu-metal** layers to isolate from Earth magnetic field
 - a **coil** inside to generate controlled inputs
 - a **3D printed structure** to located sensors and allow gradient estimates
- In order to achieve the required performance for LISA, some noise reduction techniques are applied
 - **Flipping**: applying set/reset pulses to keep magnetic moments aligned. Also removing bridge offset and drifts at low frequencies
 - **Electro-magnetic feedback**: aims to maintain bridge output near, to compensate bridge sensitivity gain due to thermal fluctuations.



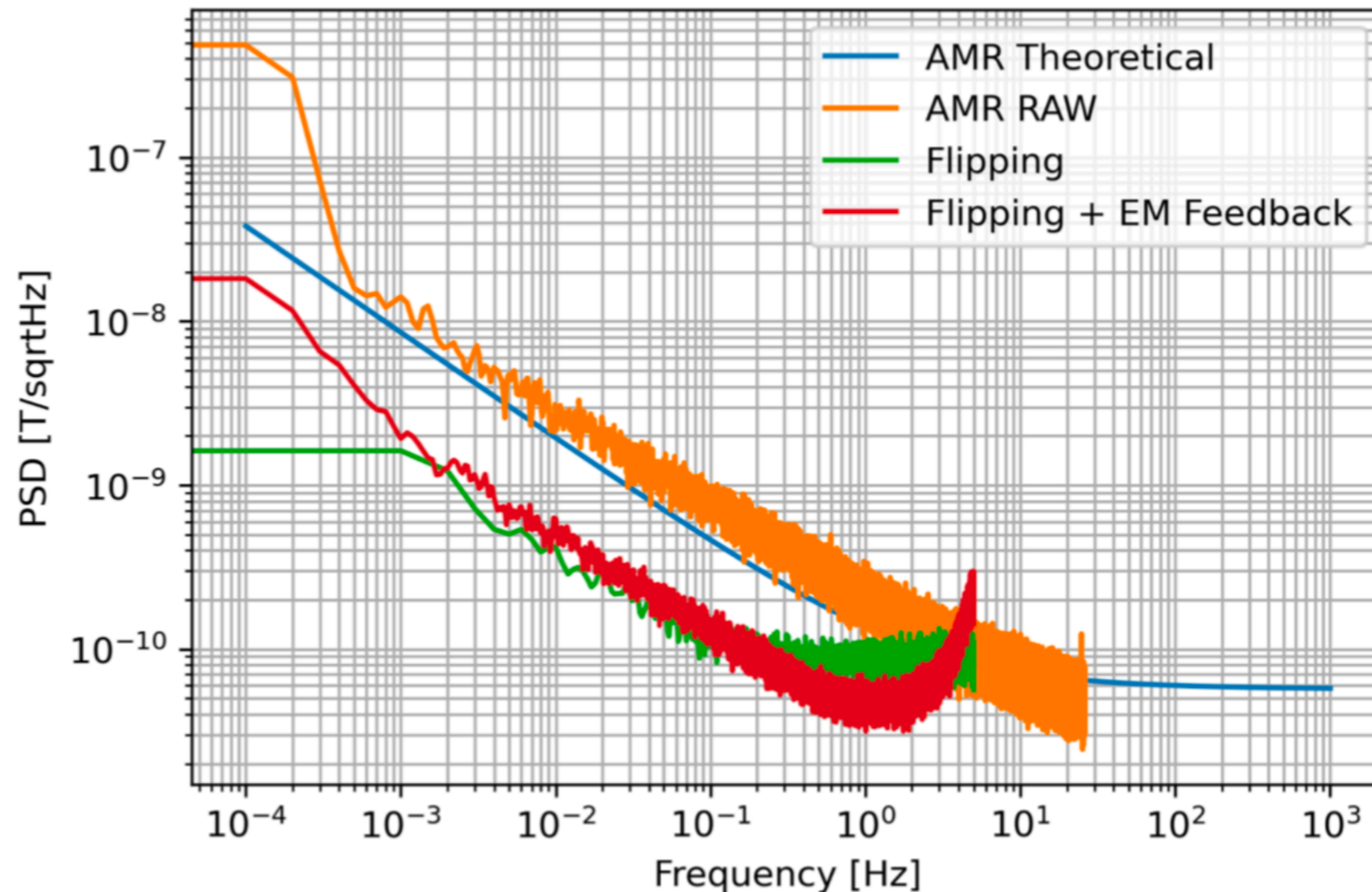
LISA – magnetic diagnostics (AMRs)



ADS1278 ADC Ev. board

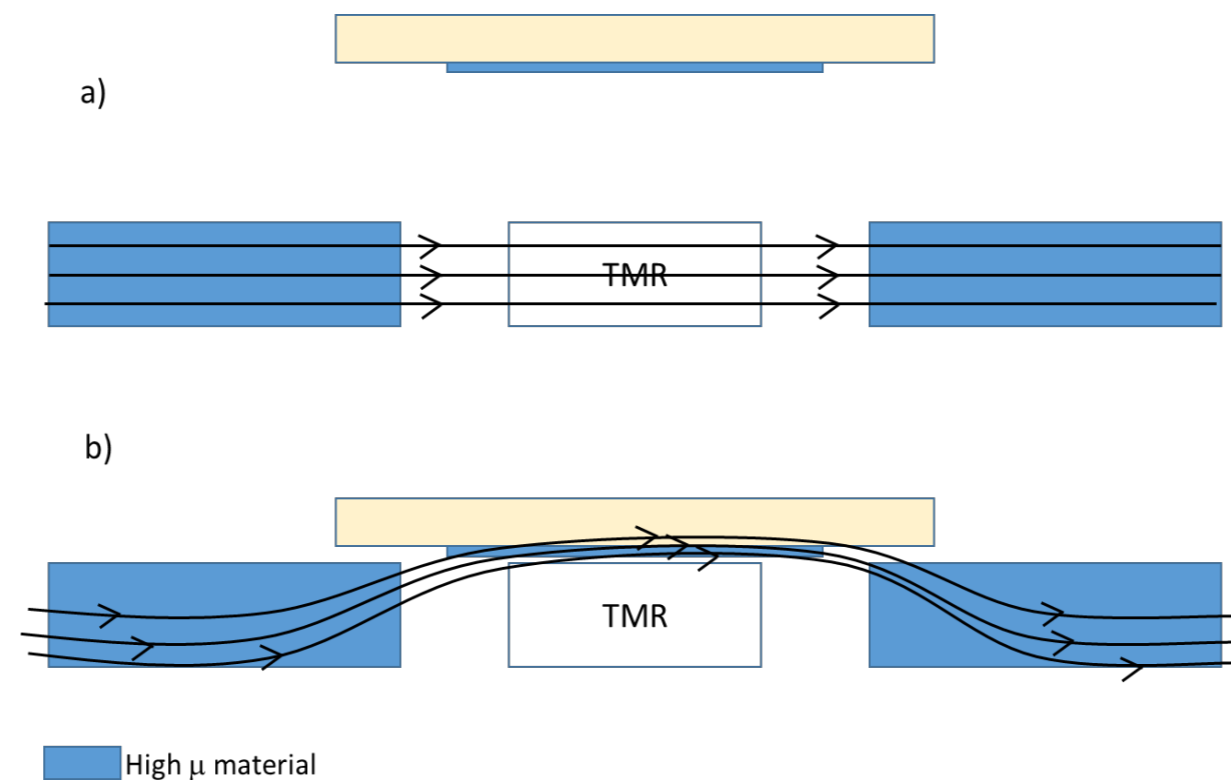
LISA – magnetic diagnostics (AMRs)

- Results show that flipping technique improves performance in the LISA band
- Similar results are obtained when applying flipping + EMF
- Currently evaluating back-action effects, i.e. magnetic field signal of a sensor as measured by another sensor



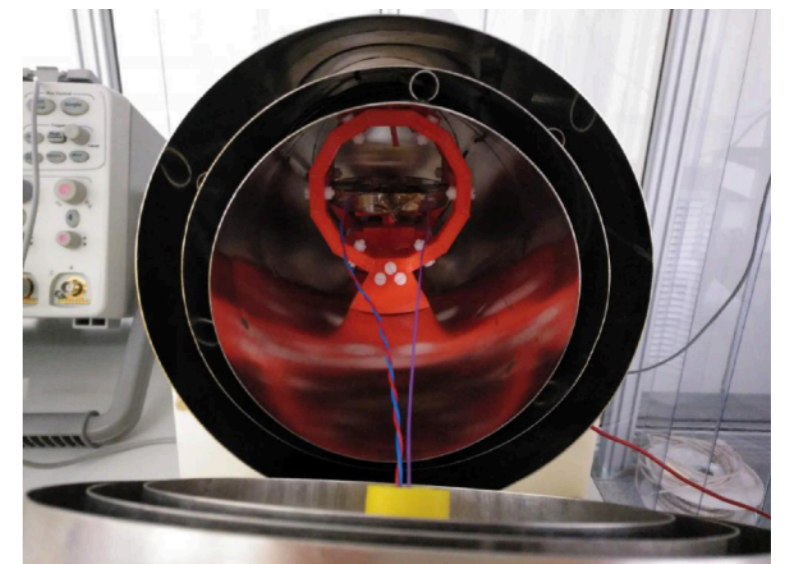
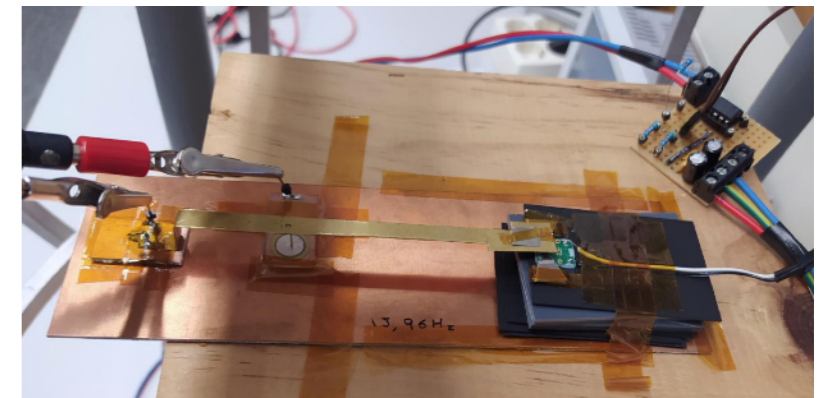
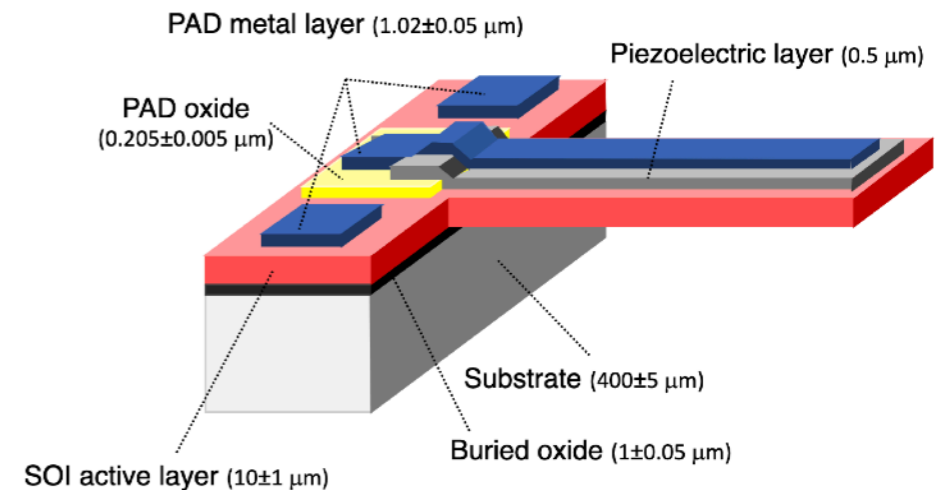
LISA – magnetic diagnostics (MEMs)

- A MEMS-based magnetometer for LISA
 - Magnetic flux is sensed using **TMR** (Tunnel Magnetoresistance)
 - Flux is enhanced using a **MFC** (Magnetic Flux Concentrator)
 - By **mechanically oscillating** the MFC near the TMRs, the Magnetic Flux is modulated at a high frequency avoiding $1/f$ noise



LISA – magnetic diagnostics (MEMs)

- Developing a compact magnetometer based on MEMs
 - Magnetic field modulation, using MEMs resonators and high permeability layers, to mitigate $1/f$ noise
 - TMR used as sensing device
- **MELISA** (*MEMS miniaturized low-noise magnetic field sensor for LISA*) got selected in an **internal IECC call** for seed funding for innovative projects across different IECC units (UPC, CSIC, UB)
- Studying resonator parameters through FEM simulations for different designs.
 - in particular resonance frequency (100-900 Hz), displacement (10-100 μm) and magnetic modulation ($< 60\%$)
- First proof-of-principle breadboard (**macrocantilever**) implemented to characterise modulation depth and read-out.
 - 10 cm long brass cantilever, reaching 1.5mm displacement at 15.7 Hz
- Characterised the noise floor of the TMR device inside mu-metal at ICE temperature control canine

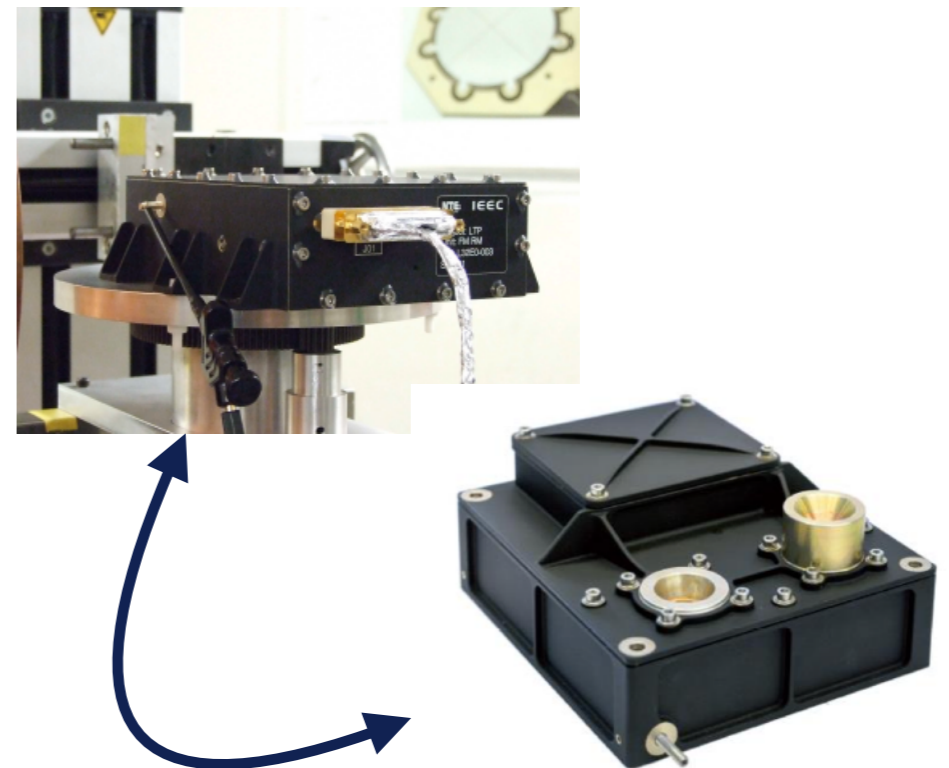


LISA - radiation monitor

- Radiation monitor to track test mass charging in the LISA test masses
 - Study the high energy radiation environment responsible for test-mass charging
 - Analyze the charge generation in test masses
 - A complete simulation **Monte Carlo model to predict direct and indirect energy deposition** will be developed.
 - Low energy particles are dominant in test mass charging
 - Adequate tools as **Penelope** or new **Geant-4** version will be used.
 - Study of radiation monitor requirements
 - ESA proposes using the standard **NGRM** in each spacecraft: is it enough?
 - Is a customized monitor needed ?- Validation of model in test beam campaigns



Armano et al. *Astroparticle Physics* 98 (2018)



LISA - radiation monitor

- Define simple Geometry in PENELOPE

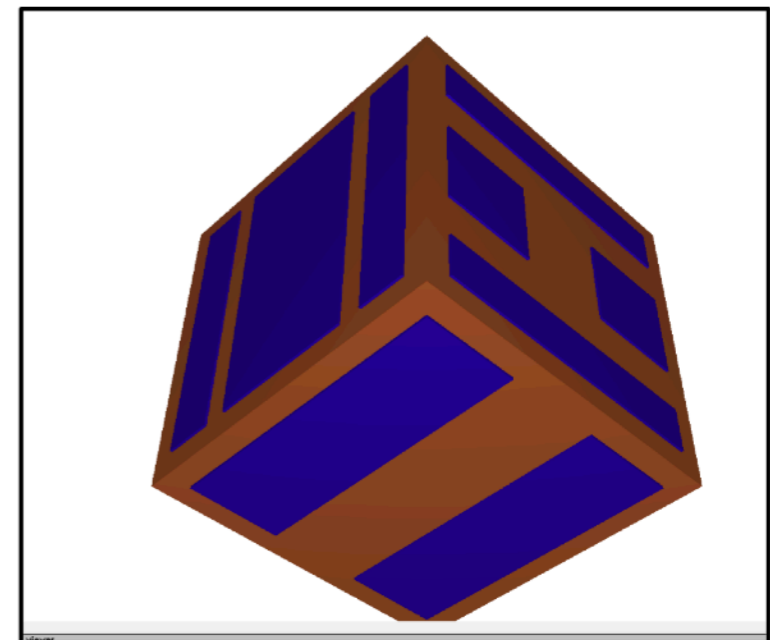
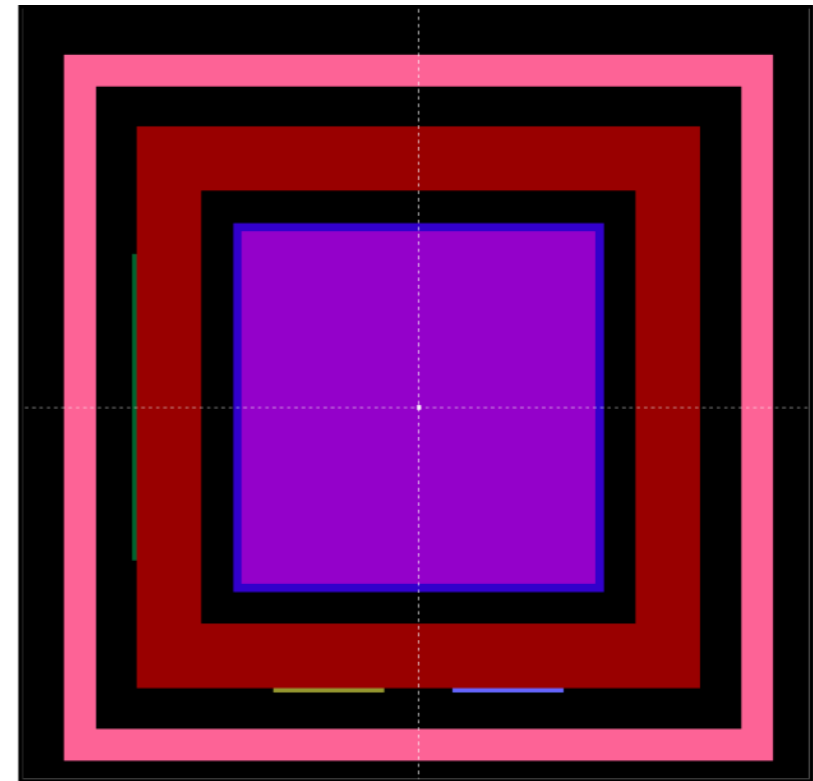
- Test Mass 46x46x46 mm³ cube
- Electrode Housing 70x70x70 mm³ cube foil
- Aluminum shielding

- Material definition:

- Al -Shielding (spacecraft)
- Pt (27%) + Au (73%) Test Mass
- Mo -Electrode Housing
- Sapphire (Al_2O_3) Electrodes

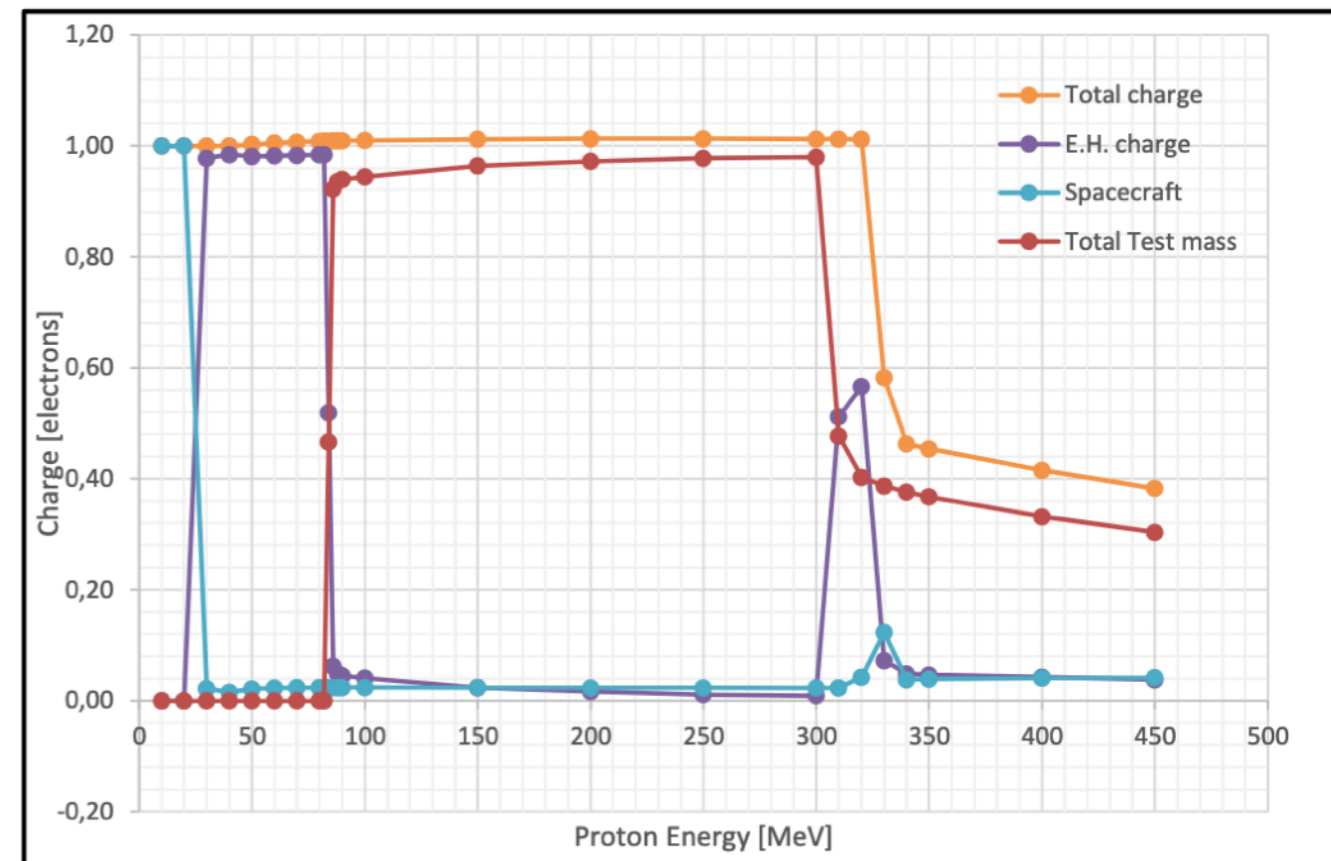
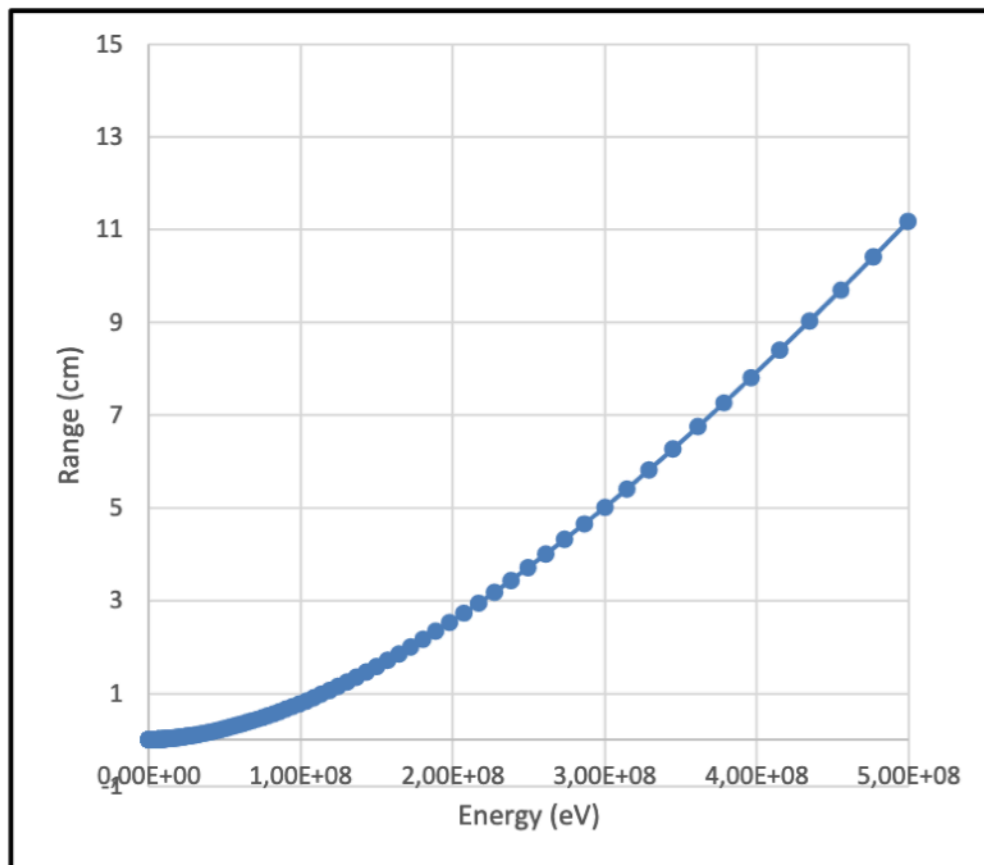
- First studies using protons

- Proton energy from 10 to 450 MeV
- Energy of absorption for the electron particle = 50 eV



LISA - radiation monitor

- First results with proton beam providing some insight
 - Around 300 MeV, proton range > test mass length
 - Test mass charge below 84 MeV is practically 0.
- Currently working in collaboration with ESA NGRM team to study in more detail the characteristics of the NGRM detector
 - ESA NGRM team providing detector response up to 1 GeV
- Next steps to consider cosmic ray spectrum and study NGRM counts to TM charge





Thanks for you attention.

Questions?